

SUPPLEMENTAL TYPE CERTIFICATE

10053107

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AIRWOLF AEROSPACE L.L.C.

**15369 MADISON ROAD
MIDDLEFIELD OH 44062
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number : US H1SW

Original Type Certificate Number : US H4SW

Type Certificate Holder : BELL HELICOPTER TEXTRON, INC.

Type Design - Model : 204B

205A-1

212

Original STC Number : FAA STC SR03408CH

Description of Design Change:

Replacement of Tension Torsion Strap Assy

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 27 April 2015


for **Massimo MAZZOLETTI**
Head of Rotorcraft Department

Note:

The following numbers are listed on the certificate:
EASA current Project Number: 0010035132-001

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Associated Technical Documentation:

ALS contained in Instructions for Continued Airworthiness AA-204-ICA, Rev. IR, 16.11.2014
or later revisions of the above listed documents approved by EASA.
Drawing B-212 TIR BAR, Rev. IR (contains Drawings AA-204-310-101/-103)
Fatigue Rep. AA-204-FaTR, Rev. IR, 26.11.2014

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- end -

Note:
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EASA current Project Number: 0010035132-001

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